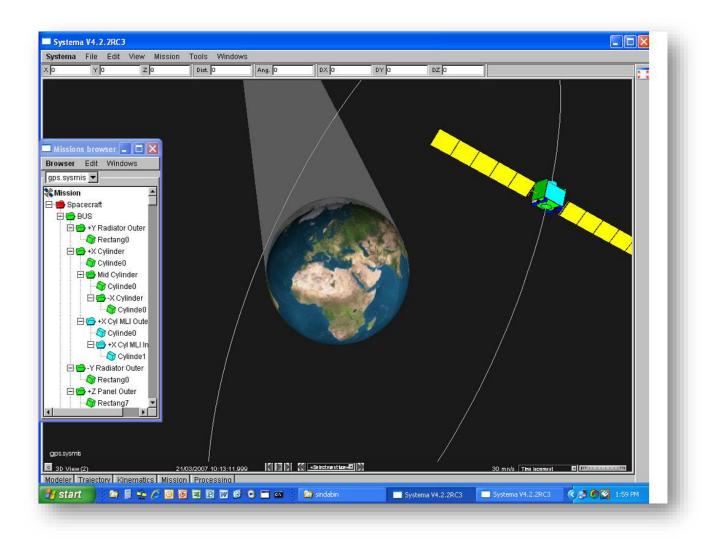
Thermica program



Orbital heat load

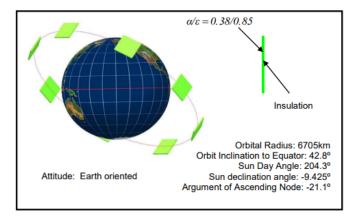
Problem 5

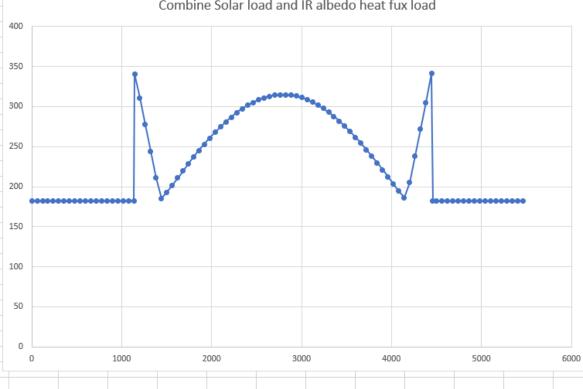
Plate in an Orbit with Eclipse

Model Description:

In this problem you will create a model like problem 4. The plate will radiate and absorb heat flux only on one side, the other side is insulated. No active heat is imposed on the plate. The orbit is a round orbit with an eclipse. The attitude is 3 axis stable, and Earth-oriented. A transient run will be made this time. Please note: this problem needs a Fortran compiler. We will add some Fortran logics in the sin file.

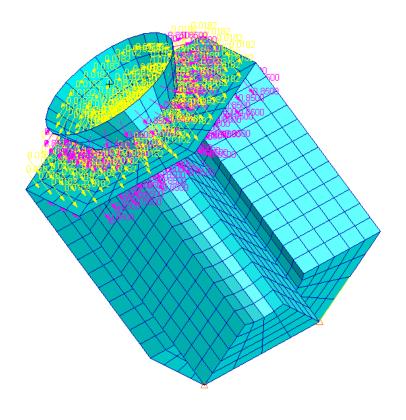
Surface Dimension = 1.0 m x 1.0 m x 0.001 m Ambient Temperature = -273.15 K The material is Aluminum 6061 T6 Thermal Conductivity = 167 W/m.K Specific Heat = 880 W/Kg Density = 2700 Kg/m³ Surface IR Emissivity = 0.85 Surface UV Absorptivity = 0.38





Combine Solar load and IR albedo heat fux load

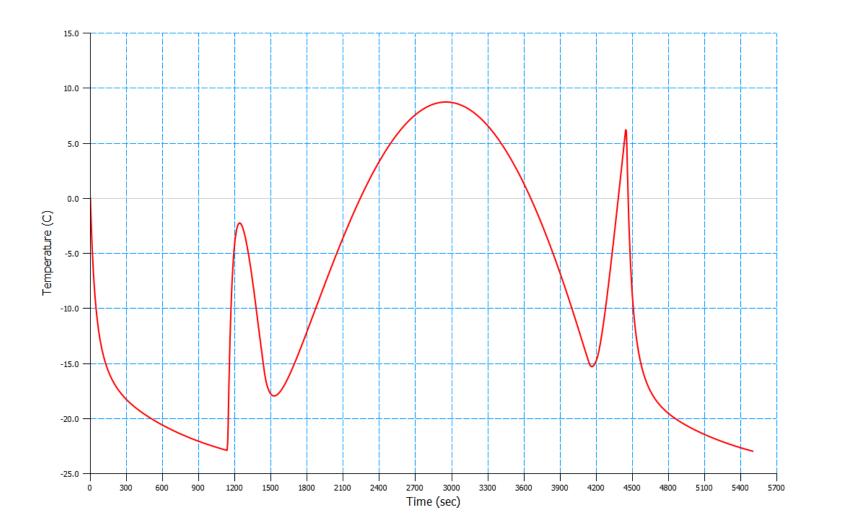
Satellite

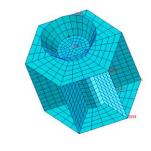


The antenna receive solar heat load ,and radiation to space at -273.15 degrees from top

Y Z

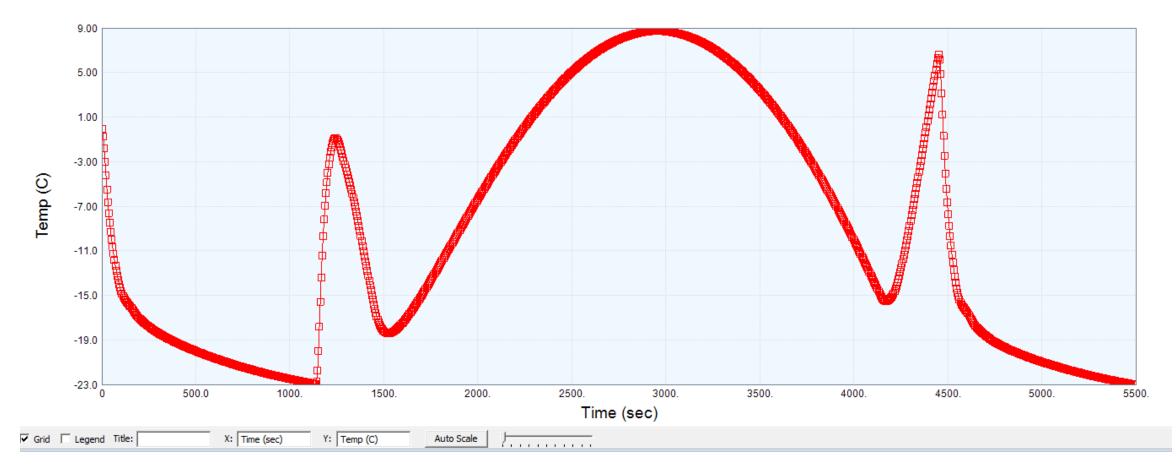
Temperature as it goes through a cycle (Node 380)





17 X

SINDAG run



As we can see that the comparison between MSC/NASTRAN thermal and SINDAG is excellent in this transient thermal analysis of satellite model, (peak temp is about 9 degree C in 9 degrees C for MSC/NASTRAN)

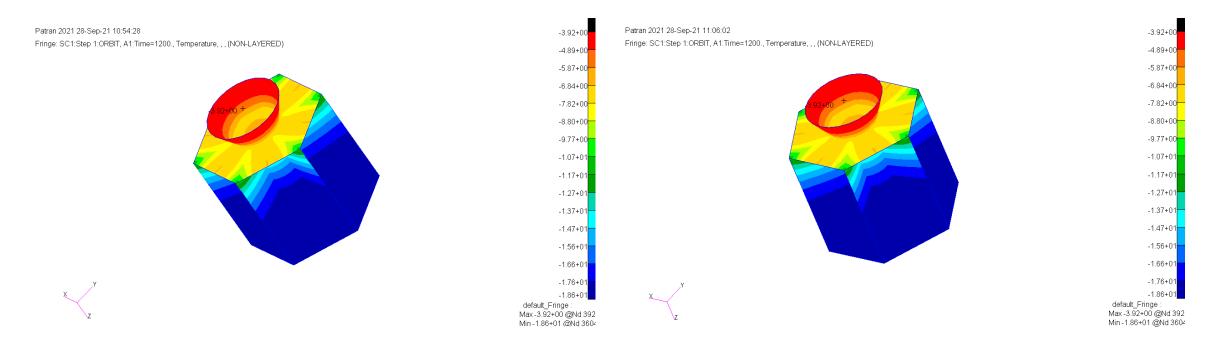
5 Thermal Conduction

© MSC Software Corporation

Temp in degree C (Time = 1200 sec))

MSC/NASTRAN

SINDAG result

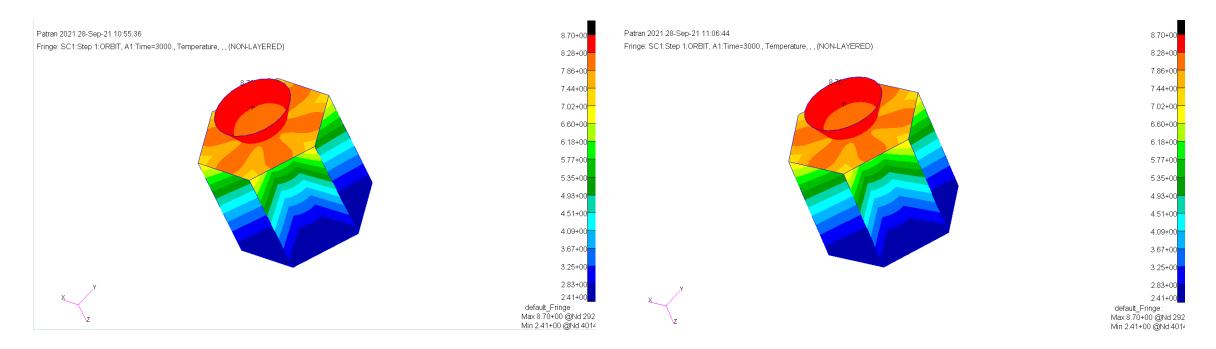


Peak temperature is -3.9 for MSC/NASTRAN and -3.9 for MSC/SINDA

(Time = 3000 sec) in the peak solar power

MSC/NASTRAN

SINDAG result

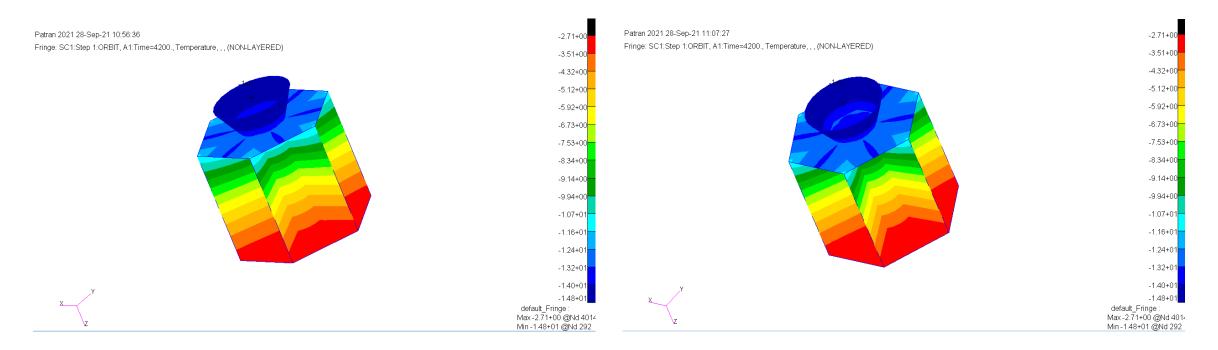


Peak temperature is 8.7 for MSC/NASTRAN and 8.7 for MSC/SINDA

(Time = 4200 sec) in the eclipse

MSC/NASTRAN

SINDAG result



Peak temperature is -2.7 for MSC/NASTRAN and 2.7 for MSC/SINDA

MSC/NASTRAN SOL 400 transient thermal analysis

\$ Watt/c SOL 400	m2.k4							
CEND								
<pre>\$ Direct</pre>	Text Ir	nput for	Global	Case Co	ontrol Data	a –		
ANALYSIS	= HTRAM	l						
TITLE =	MSC/NAST	Fran Job	CREATED) ON 25-	SEP-98 AT	21:21:	01	
ECHO = N	IONE							
SET 77 =	5001							
IC = 1								
SUBCASE								
SUBTI	TLE=trar	n_new						
TSTEP	NL = 1							
DLOAD	= 2							
SPC=1								
THERM	IAL(SORT2	2,PRINT)	=ALL					
SPCF=77								
<pre>\$ Direct</pre>		nput for	this Su	ibcase				
OUTPUT(X								
xgrid=ye								
ygrid=ye								
xtitle=t								
ytitle=t								
xyplot t		1(T1),38	0(T1)					
BEGIN BU		_						
<pre>\$ Direct</pre>			Bulk Da	ata				
MDLPRM	HDF5	0						
PARAM	TABS	273.15						
PARAM, SI								
PARAM	PRGPST	NO	-					
TSTEPNL		1100	5.	1	ADAPT	2	-10	U
	.01	•						
e		0						
<pre>\$ Dynami TABLED1</pre>		able :	solar_10	bad_new				
*	10 0.		404 67	2000	60.		101 (7	7000
*	0. 120.		181.67		180.		181.67	
*			181.67				181.67	
*	240.		181.67		300.		181.67	
*	360. 480.		181.67		420. 540.		181.67	
*	480.		181.67		540. 660.		181.67	
*	720.		181.67		780.		181.67	
*	720. 840.							
*	840. 960.		181.67		900. 1020.		181.67	
-								
*	1000							
*	1080. 1145.		181.67 340.67		1136. 1200.		181.67 310.64	

			of Load	Set :	view_bottom	
SPC,1,5001,1,-273.15						
temp,1,50						
\$VIEW	2	101	BOTH			
\$RADSET	101					
TLOAD1	5	3			10	
\$LOAD1	6	4			10	
\$LOAD	2	1.	1.	5	1.	
DLOAD	2	1.	1.	5		
RADBC, 500	01,0.5,,	100001,†	thru,100)552		
<pre>\$ Normal</pre>	Heat Fl	ux of Lo	oad Set	: anter	nna_top	
QBDY3	3	1e-4		100	001	
QBDY3	3	1e-4		100	202	
QBDY3	3	1e-4		100	003	
QBDY3	3	1e-4		100	004	
QBDY3	3	1e-4		100	005	
QBDY3	3	1e-4		100	206	
QBDY3	3	1e-4		100	007	
QBDY3	3	1e-4		100	208	
QBDY3	3	1e-4		100	209	
QBDY3	3	1e-4		100	010	
QBDY3	3	1e-4		100	011	
QBDY3	3	1e-4		100	012	
QBDY3	3	1e-4		100	013	
QBDY3	3	1e-4		100	014	
QBDY3	3	1e-4		100	015	
QBDY3	3	1e-4		100	016	

6

MSC/SINDAG test deck

SC Sinda input file created for problem name: satellite rcnt	-241153, 4263, 4286, 4.692278
BCD 3THERMAL LPCS	-241154, 4264, 4286, 2.166580
BCD 9MSC Sinda model from Patran/Nastran and SindaRAD	-241155, 4265, 4286, 4.060300
BCD 9 Model name = satellite rcnt	-241156, 4266, 4286, 5.488096
END	-241157, 4278, 4286, 5.504487
BCD 3NODE DATA	-241158, 4279, 4286, 4.095987
282, 100.0000 , 0.5141574	-241159, 4281, 4286, 4.738801
283, 100.0000 , 0.6089281	-241160, 4282, 4286, 4.692272
284, 100.0000 , 0.3351715	-241161, 4283, 4286, 2.166581
285, 100.0000 , 0.6837769	-241162, 4284, 4286, 4.060296
286, 100.0000 , 0.5147809	-241163, 4285, 4286, 5.488082
287, 100.0000 , 0.6090962	END
288, 100.0000 , 0.3346427	BCD 3CONSTANTS DATA
289, 100.0000 , 0.6699428	DRLXCA=0.1000000E-02
290, 100.0000 , 0.5152241	ARLXCA=0.1000000E-02
291, 100.0000 , 0.6091574	NLCOP=5000
292, 100.0000 , 0.3342964	OUTPUT=5.000000
293, 100.0000 , 0.6840760	TIMEND=5500.000
294, 100.0000 , 0.5147811	DTIME1=5.000000
295, 100.0000 , 0.6090990	GRVCON=9.810000
296, 100.0000 , 0.3346427	SIGHA=0.5670000E-11
298, 100.0000 , 0.5141553	TMP2R0=0.000000
299, 100.0000 , 0.6089281	END
300, 100.0000 , 0.3351700	BCD 3ARRAY DATA arrays from finite element model functions
305, 100.0000 , 0.6837753	10 \$
306, 100.0000 , 0.5147801	0.000000 , 181.6800 , 60.00000 , 181.6800
307, 100.0000 , 0.6090965	
308, 100.0000 , 0.3346436	
309, 100.0000 , 0.6699433	
310, 100.0000 , 0.5152220	400,0000 101,0000 540,0000 101,000
311, 100.0000 , 0.6091556	500 0000 101 5000 550 0000 101 5000
312, 100.0000 , 0.3342972	700 0000 101 C000 700 0000 101 C000
313, 100.0000 , 0.6840753	
313, 100.0000 , 0.8840753 314, 100.0000 , 0.5147810 315, 100.0000 , 0.6090971	960.0000 , 181.6800 , 1020.000 , 181.68